

# Application of Harmonic Balance Method for Non-linear Gust Responses

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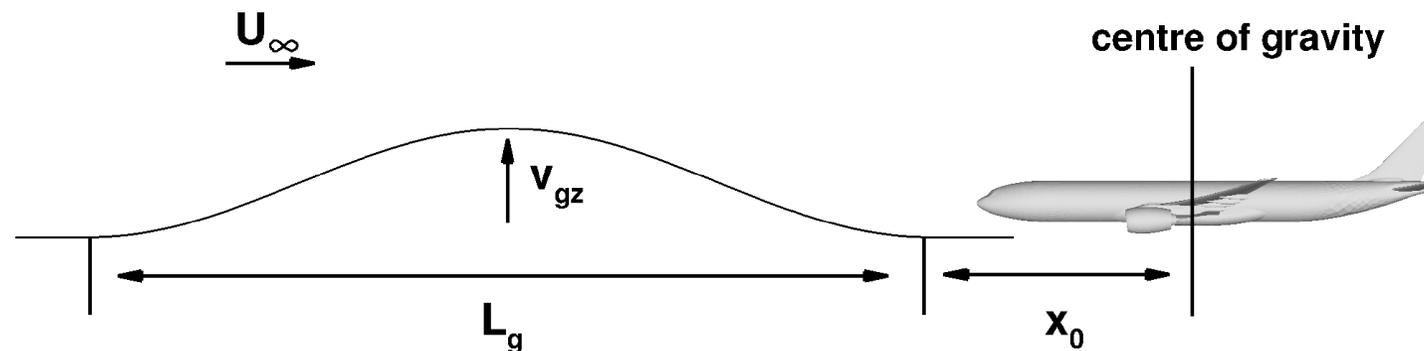
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SciTech – Structural Dynamics

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## Motivation

- Gust analysis one challenge in certification
- Covering a large parameter space
- Linear potential methods (DLM) fail in transonic regime
- Non-linear RANS equations coupled to structure and flight dynamics computationally too expensive
- Linearised RANS methods retain RANS accuracy at significantly reduced cost



## Linearised Frequency-Domain: A Short Introduction

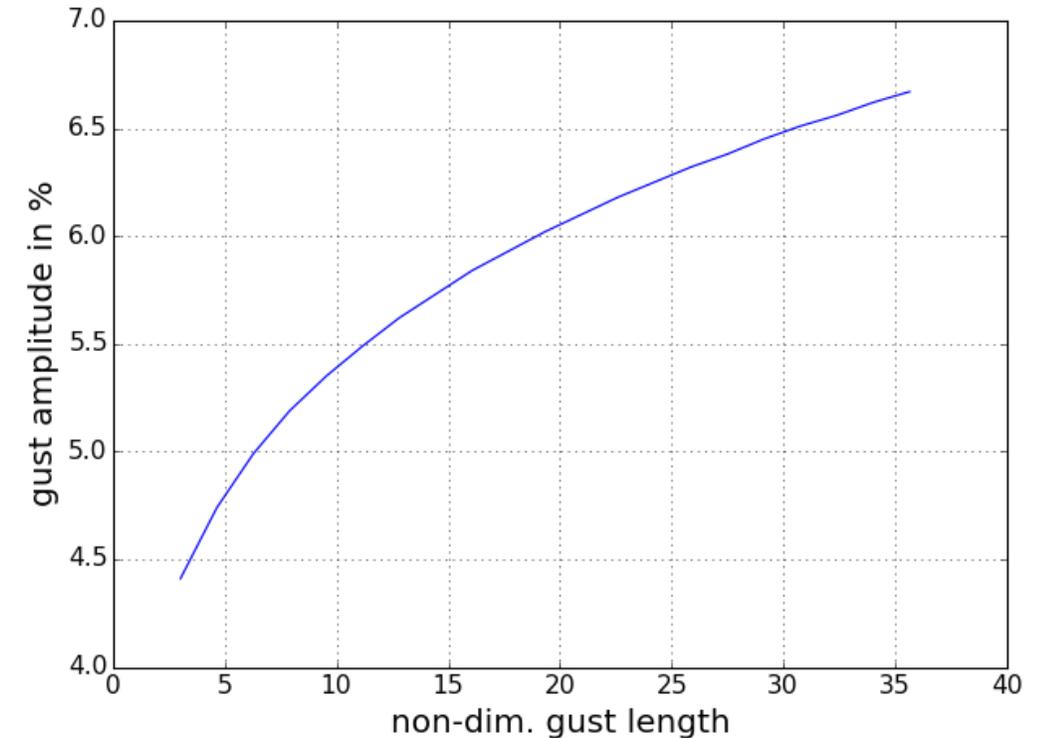
- Starting with spatially discretised RANS equations
- Separate variables in steady mean state and small time-dependent perturbation
- Linearise non-linear residual function around steady flow-field
- Transform equation into frequency domain
- Obtain a large, but sparse system of linear equations

$$\left( \frac{\partial R}{\partial W} - j\omega I \right) \hat{W} = - \frac{\partial R}{\partial v_g} \hat{v}_g$$

# Frequency-Domain Non-linear Gust Response Computation

## *Motivation*

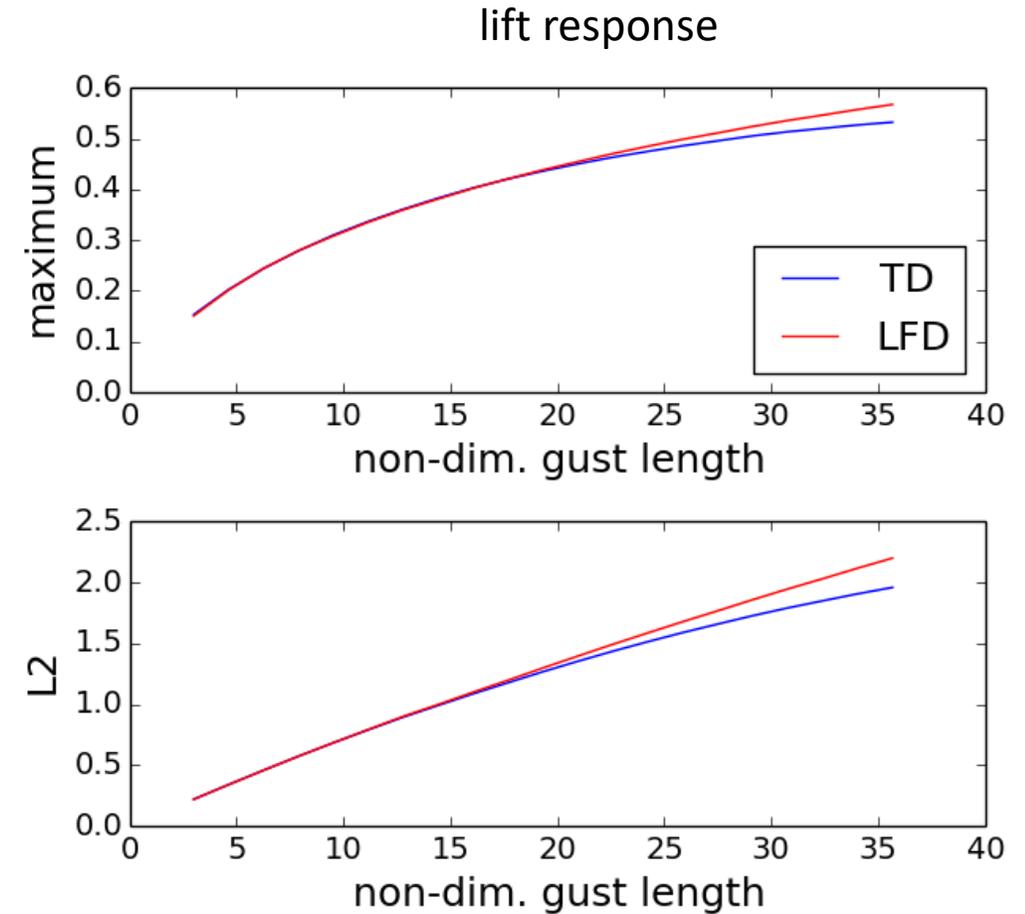
- CS 25: gust amplitude increases with gust length
- Linearised frequency domain (LFD) accurate for infinitesimally small amplitudes
- Impact on accuracy considering certification amplitudes ?



# Frequency-Domain Non-linear Gust Response Computation

## Motivation

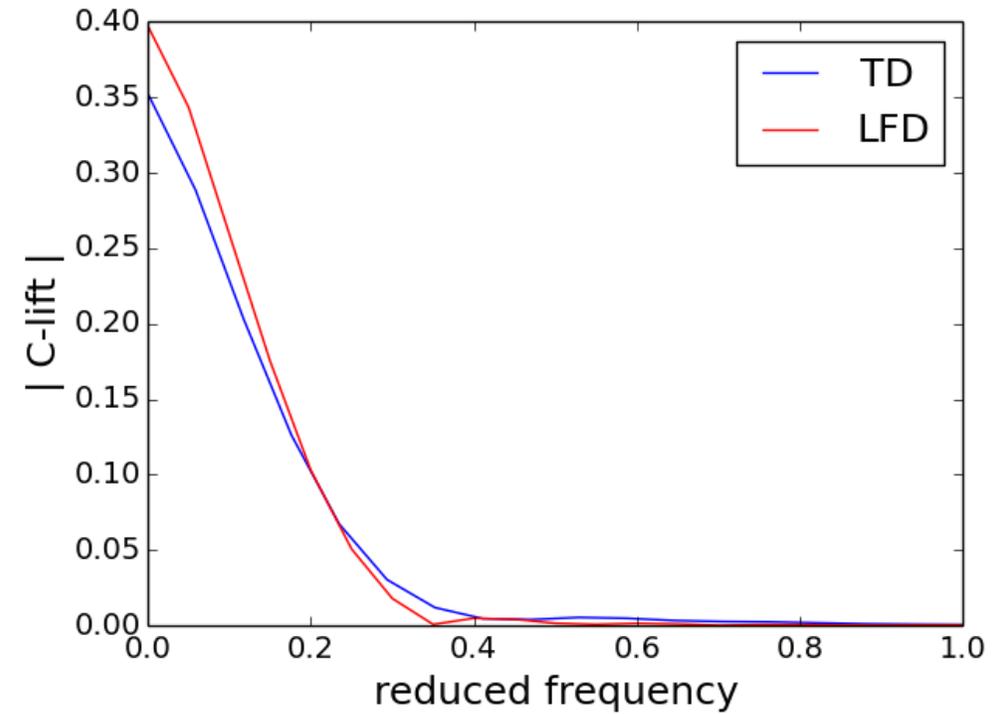
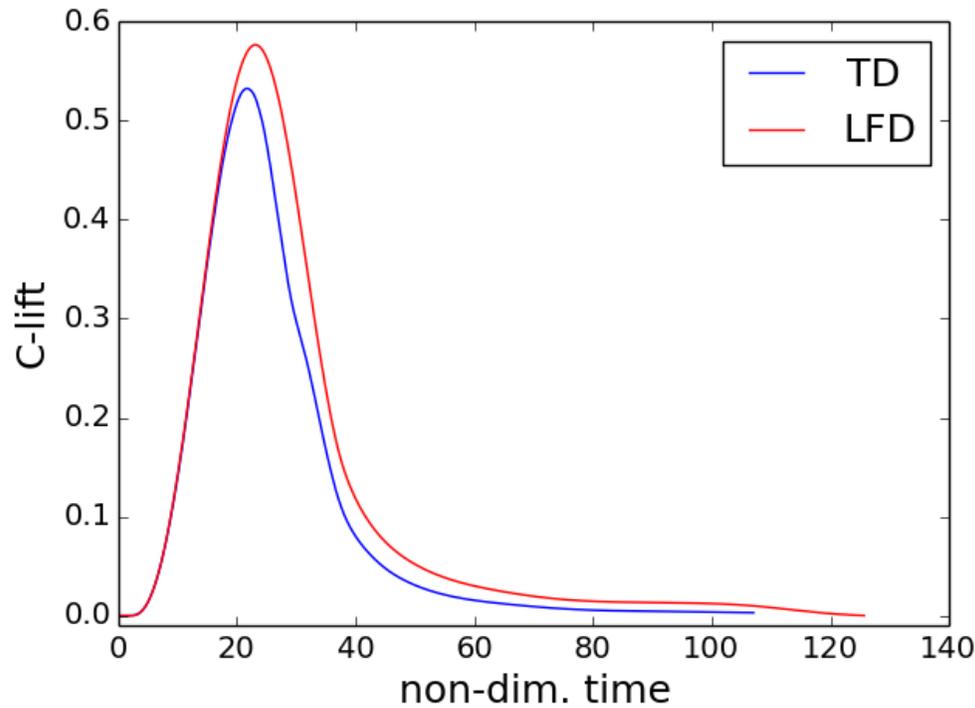
- CS 25: gust amplitude increases with gust length
- Linearised frequency domain (LFD) accurate for infinitesimally small amplitudes
- Impact on accuracy considering certification amplitudes?
  - Compare LFD to non-linear time-domain simulations
  - Shown is max. lift response and  $\int_0^T \Delta c_L^2$  for NACA0012 test case
  - Good agreement till non-dim. gust length 20 (for a typical aircraft case: about 120m)



# Frequency-Domain Non-linear Gust Response Computation

## Motivation

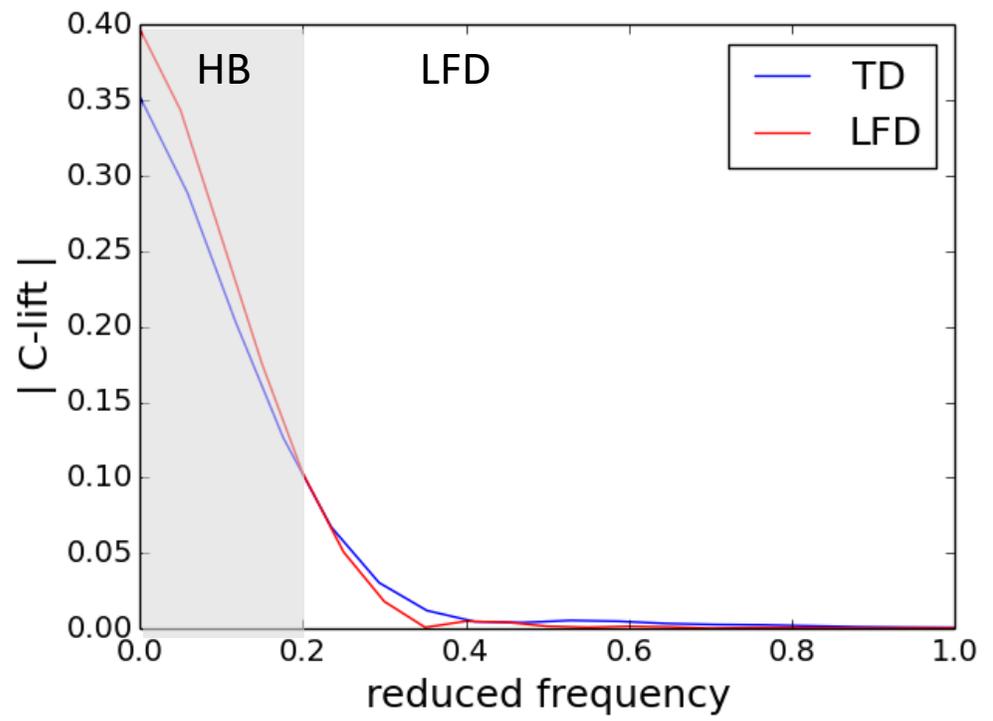
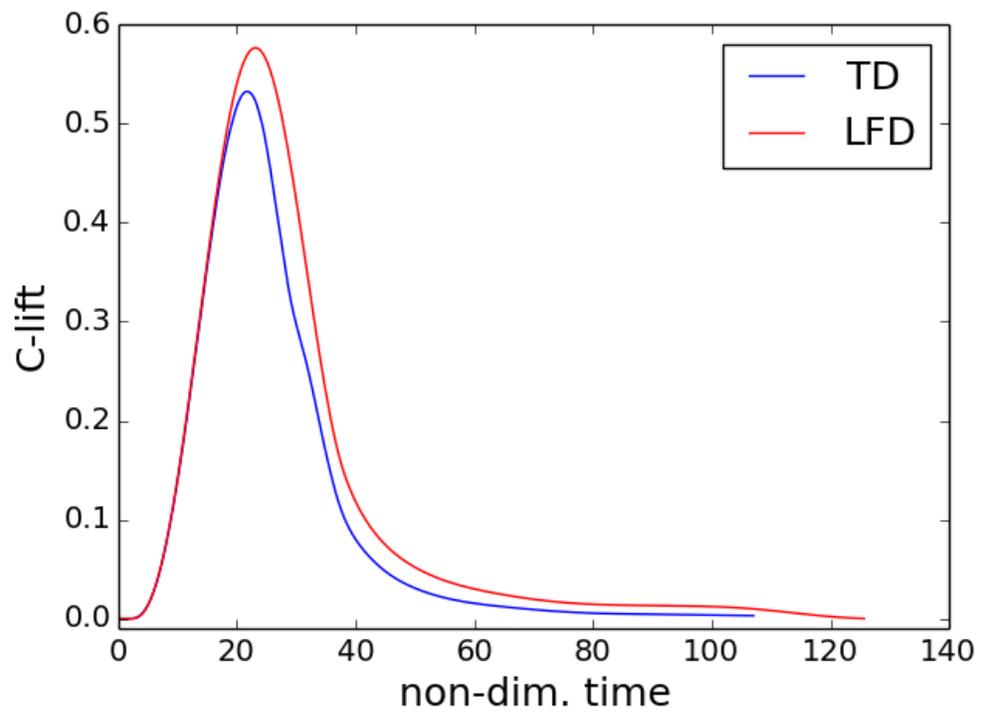
- Detailed analysis of largest gust length of 35



# Frequency-Domain Non-linear Gust Response Computation

## Idea

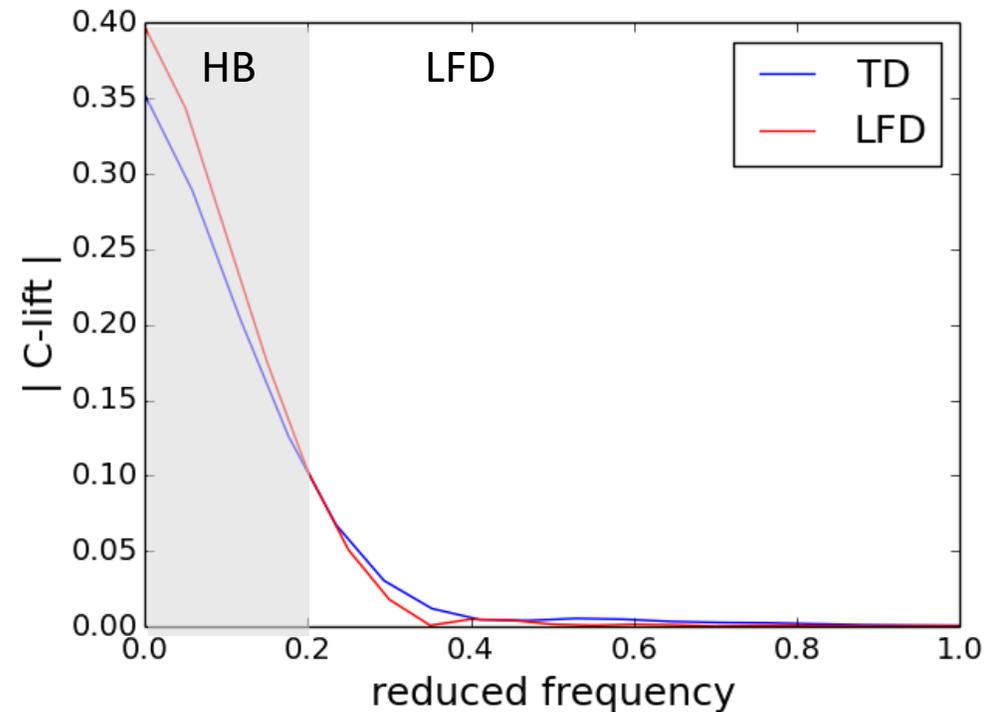
- Use Harmonic Balance (HB) method to enhance accuracy at low frequencies



# Frequency-Domain Non-linear Gust Response Computation

## Idea

- Use Harmonic Balance (HB) method to enhance accuracy at low frequencies
- Not an amplitude non-linearity "per frequency"  
(minor effect)
- Reduction in magnitude due to coupling between the harmonics of excitation and response  
→ HB must be used for 1-cos gust,  
not single frequency sinusoidal
- Observation made for an aerofoil, but can we see a similar result for an aircraft case?

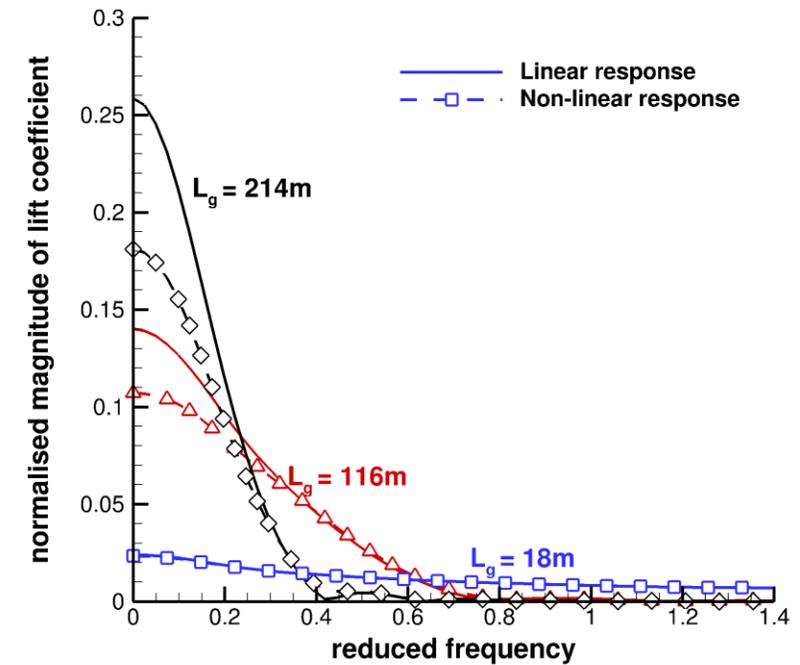
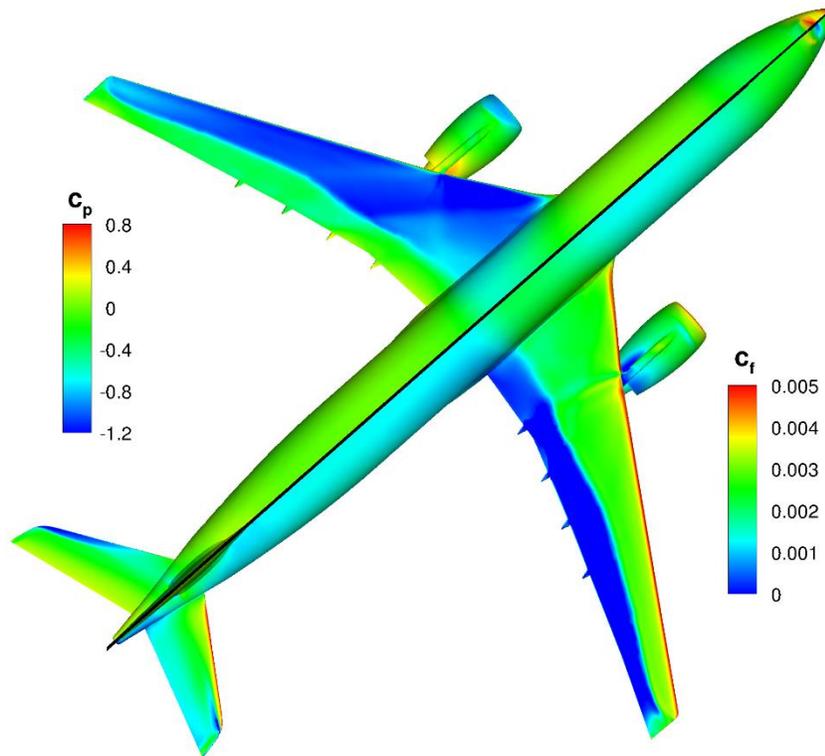


# Frequency-Domain Non-linear Gust Response Computation

## Idea

same for full aircraft case at cruise flight

Instantaneous  
at  $c_{L-max}$



# Frequency-Domain Non-linear Gust Response Computation

## *Approach*

1. Calculate steady-state solution
2. Compute LFD solutions covering the relevant frequency range
3. Reconstruct time-domain response for small and medium gust lengths
4. For each “non-linear” gust length:
  1. Choose a base frequency and number of harmonics for Harmonic Balance method
  2. Solve HB equation
  3. Add LFD solutions for frequencies that are not covered by HB
  4. Reconstruct time-domain response

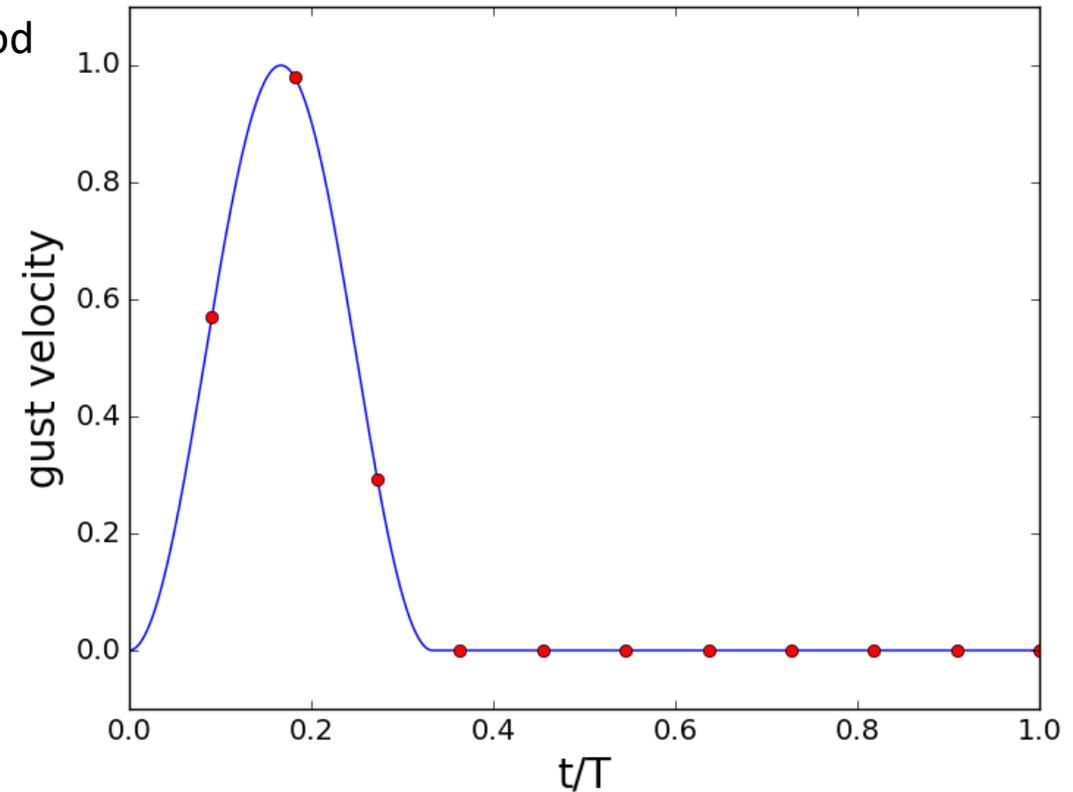
# Frequency-Domain Non-linear Gust Response Computation

## HB Approach

1.  $2N_H + 1$  solution vectors equidistantly distributed over a period
2. Compute at each time-slice the residual vector
3. Transform into frequency-domain
4. Compute update via pseudo-time integration

$$\frac{dW_{HB}}{d\tau} = \omega_b DW_{HB} + R_{HB}$$

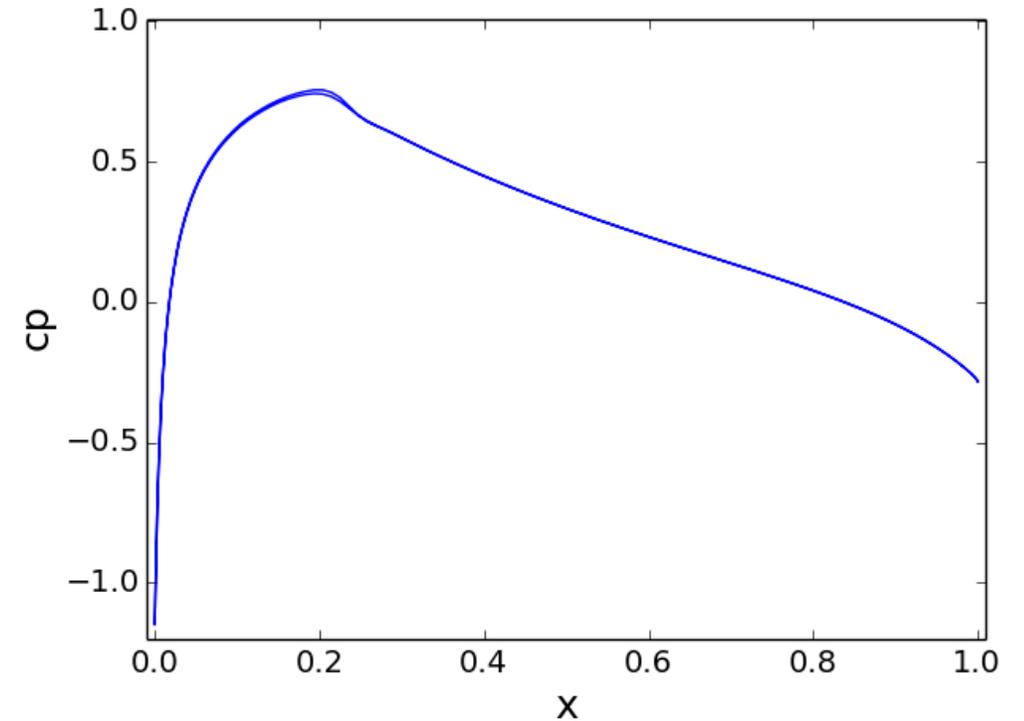
$$D_{ik} = \frac{2}{2N_H - 1} \sum_{m=1}^{N_H} m \sin \frac{2\pi(k-i)m}{2N_H + 1}$$



## Frequency-Domain Non-linear Gust Response Computation

*Results: NACA0012*

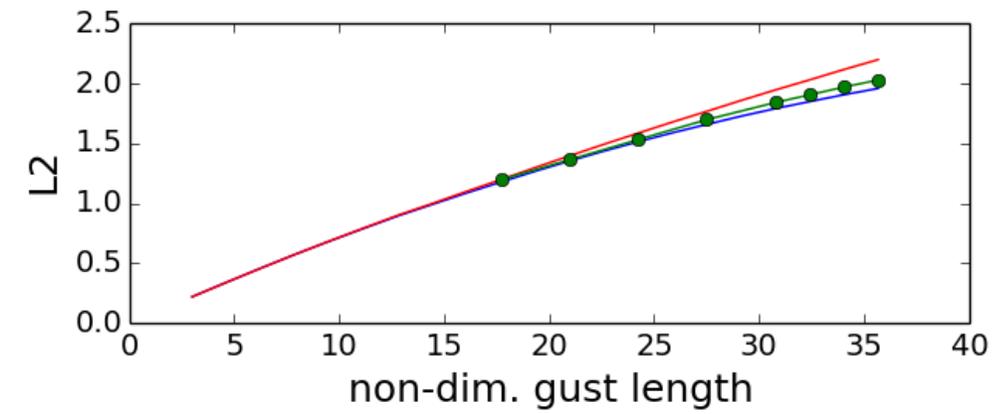
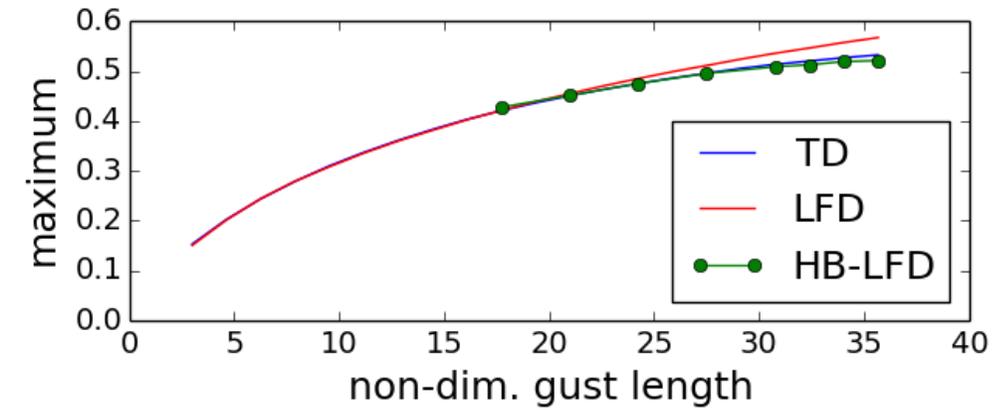
- Mach 0.75, AoA = 0 deg., Re = 10 million
- Weak transonic case



# Frequency-Domain Non-linear Gust Response Computation

*Results: NACA0012 – HB-LFD with 3 harmonics*

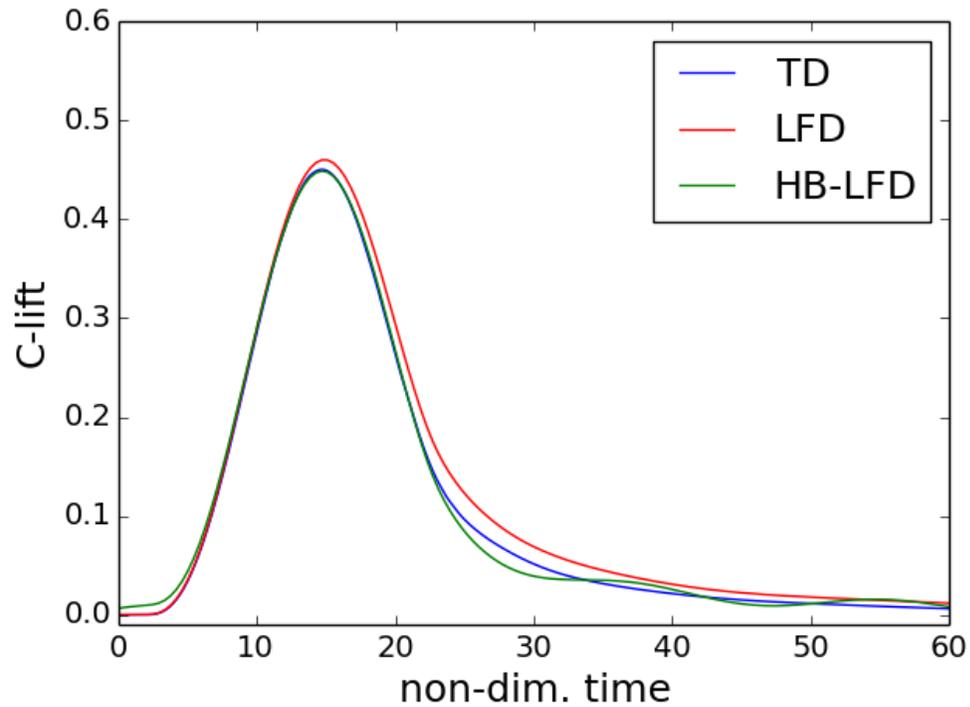
- Mach 0.75, AoA = 0 deg., Re = 10 million
- Weak transonic case
- Harmonic Balance with 3 harmonics
- Significant improvement in both norms
- Small deviations remain at highest gust lengths
- about 5x faster than TD per gust simulation



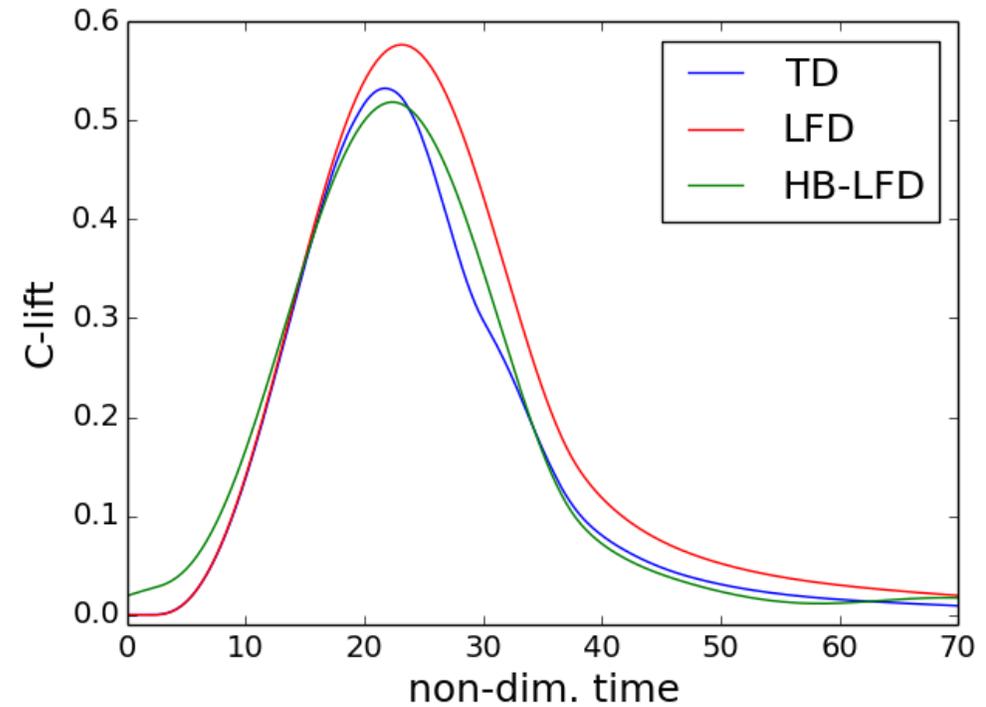
# Frequency-Domain Non-linear Gust Response Computation

*Results: NACA0012 – HB-LFD with 3 harmonics*

Gust length = 21



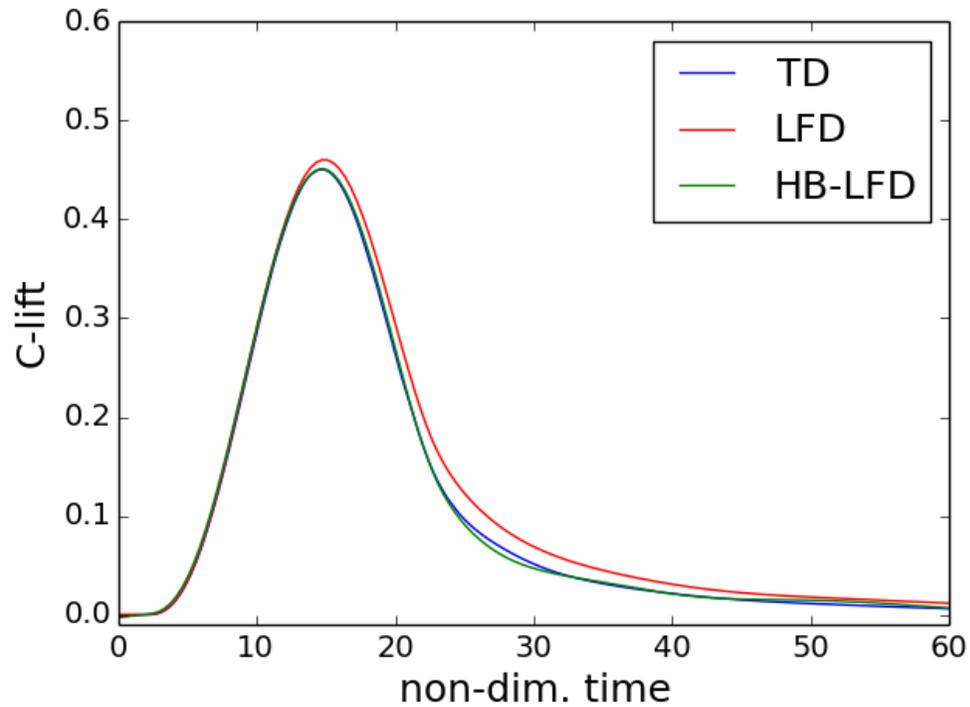
Gust length = 35



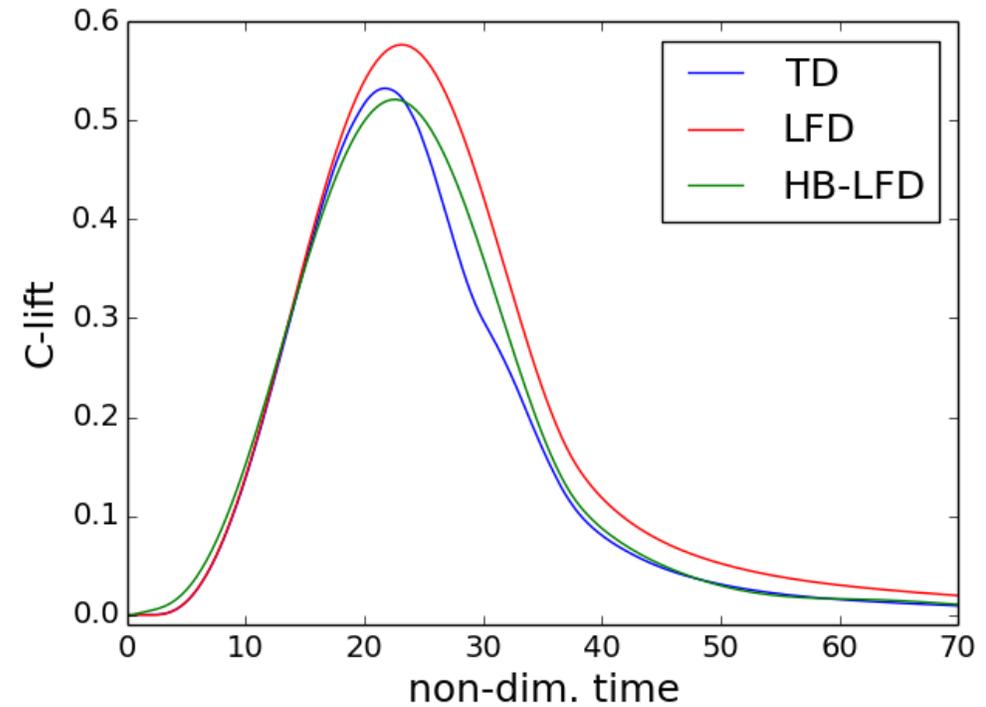
# Frequency-Domain Non-linear Gust Response Computation

*Results: NACA0012 – HB-LFD with 4 harmonics*

Gust length = 21



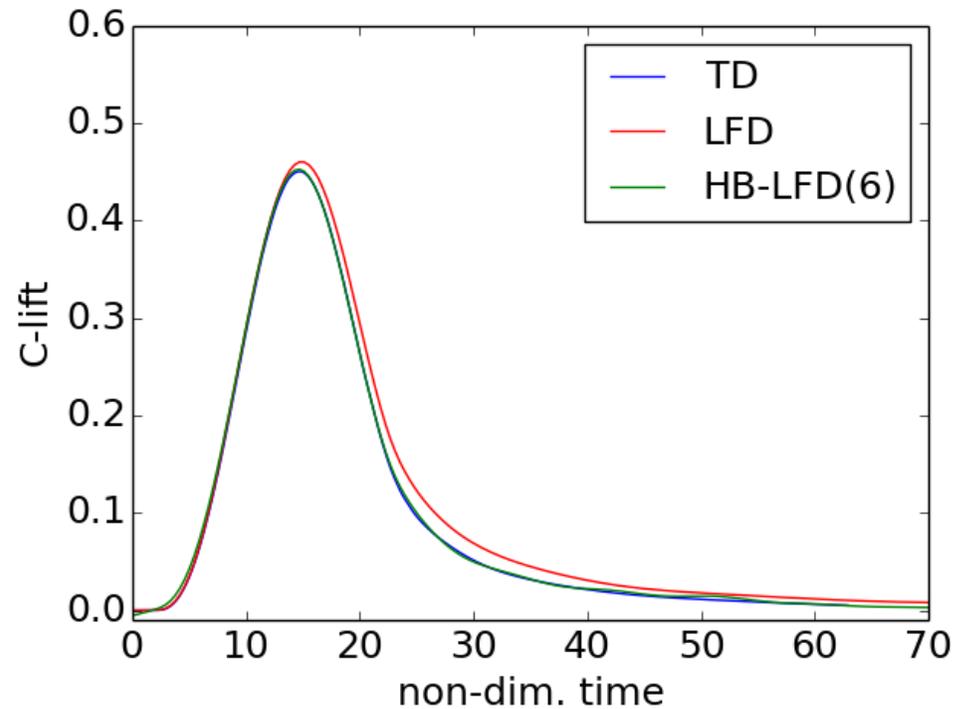
Gust length = 35



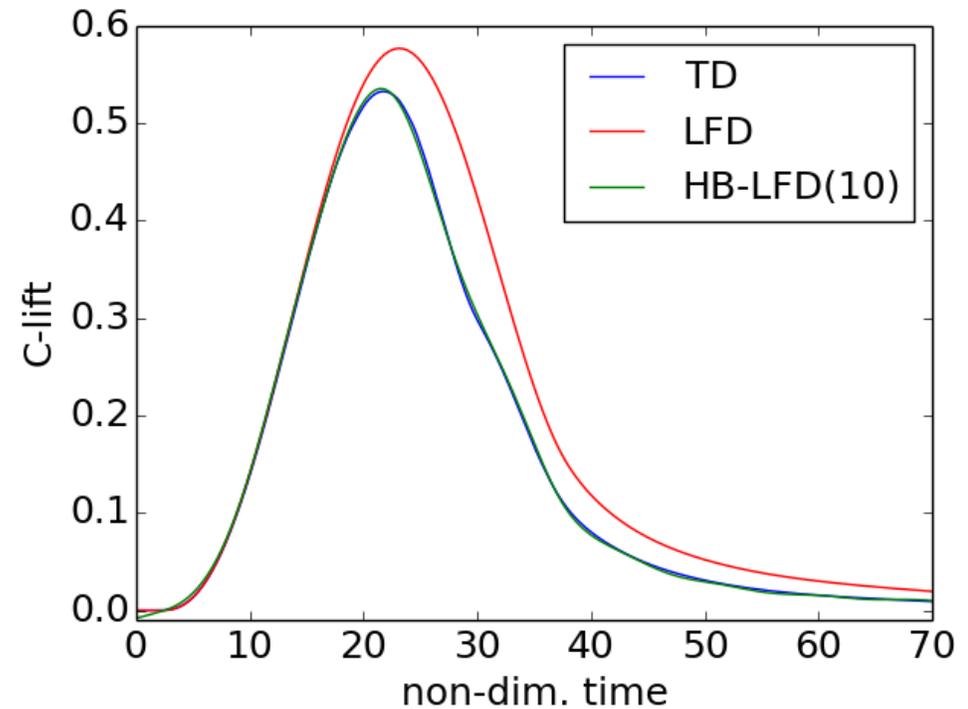
# Frequency-Domain Non-linear Gust Response Computation

*Results: NACA0012 – HB-LFD best fit*

Gust length = 21, 6 harmonics



Gust length = 35, 10 harmonics



## Intermediate conclusion

- Aerodynamic responses of gust encounter compared between linearised frequency domain and non-linear time-domain simulations using CS-25 gust definitions
  - Good agreement for small and medium gust lengths for NACA0012 aerofoil
  - Lift response over-estimated by LFD for larger gust lengths and amplitudes
- Applying Harmonic Balance method with a small number of harmonics combined with LFD results for higher frequencies yields improvement for NACA0012

### Next step:

- Compute gust response of fluid-structure coupled configuration using Harmonic Balance and LFD

## LFD4Gust with FSI

- Rearrange structural equation in system of 1<sup>st</sup> order ODE
- Augmented LFD system

$$\left( \begin{bmatrix} A_{ff} & A_{fs} \\ A_{sf} & A_{ss} \end{bmatrix} - j\omega I \right) \begin{bmatrix} w_f \\ w_s \end{bmatrix} = \begin{bmatrix} b_f \\ 0 \end{bmatrix}$$

with subscripts  $f$  and  $s$  denoting fluid or structural DoF, respectively

- Right-hand-side vector defined by field-velocity method

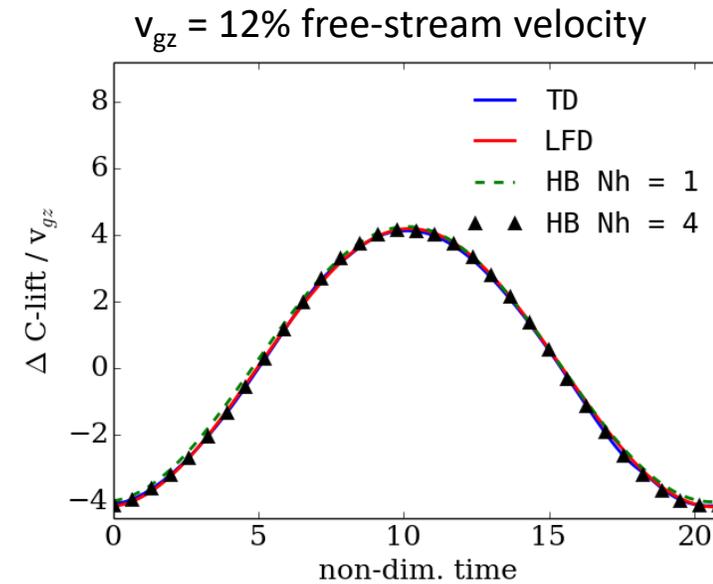
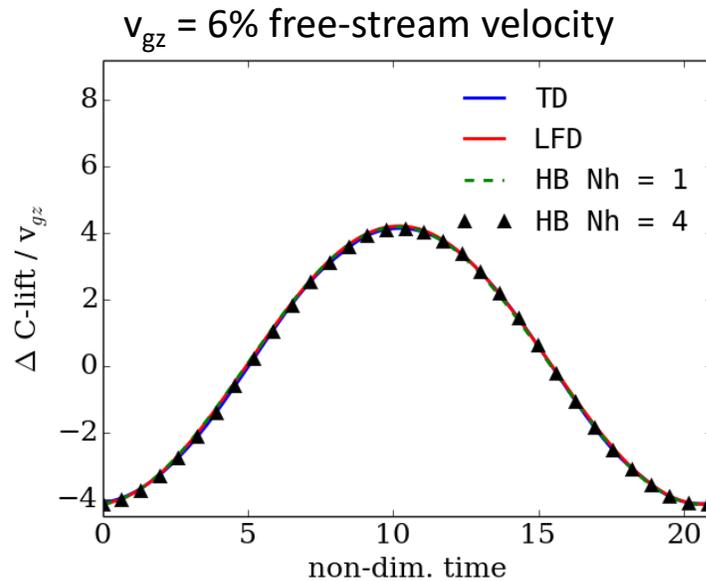
## HB4Gust with FSI

- Similar to LFD, the system of equations and the vector of unknowns is augmented with their structural part
- Thus, HB solves for  $W_f$  and  $W_s$  at each time slice
- Corresponding fluid and structural residuals are computed
  - Involves updating grid point locations and velocities according to structural motion for each time slice
  - Grid movement can be realised using deformation or here rigid-body motion (pitch-plunge aerofoil)
- The rest is usual HB approach
- For implicit solution scheme, coupled Jacobians are used (see LFD solver)

## HB4Gust with FSI

- Previous test case extended by pitch-plunge structure
- In-vacuum reduced frequencies of 0.34 for heave and 1.0 for pitch
- Sinusoidal gust encounter with wave length of 21 chord lengths and two gust amplitudes
- TD Signal recorded after 20 periods

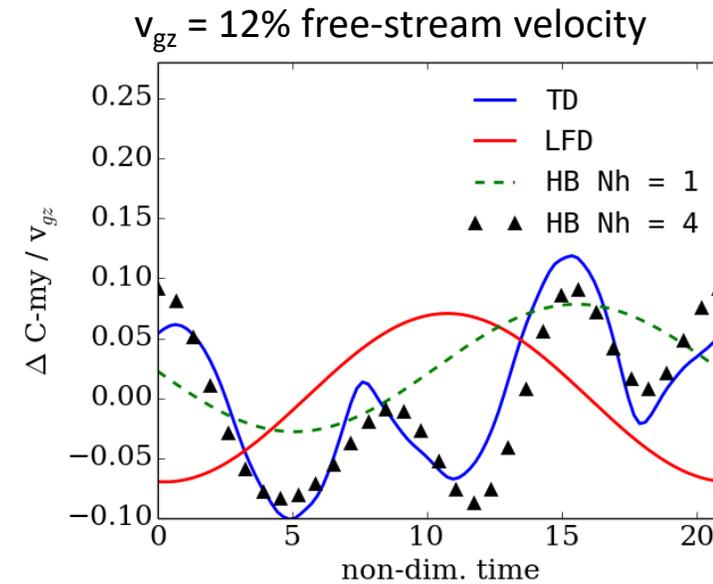
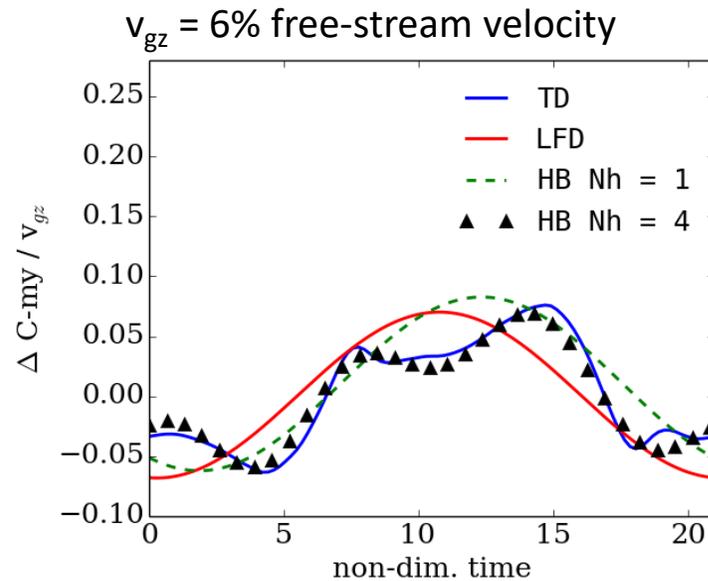
Lift  
response



## HB4Gust with FSI

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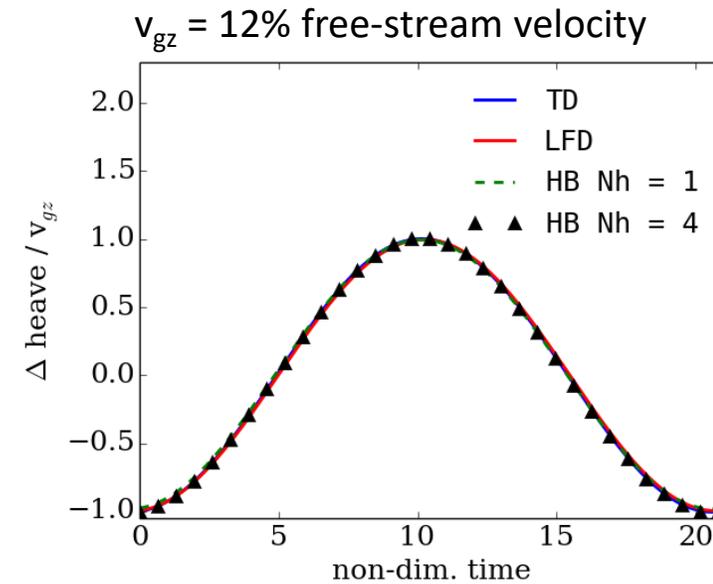
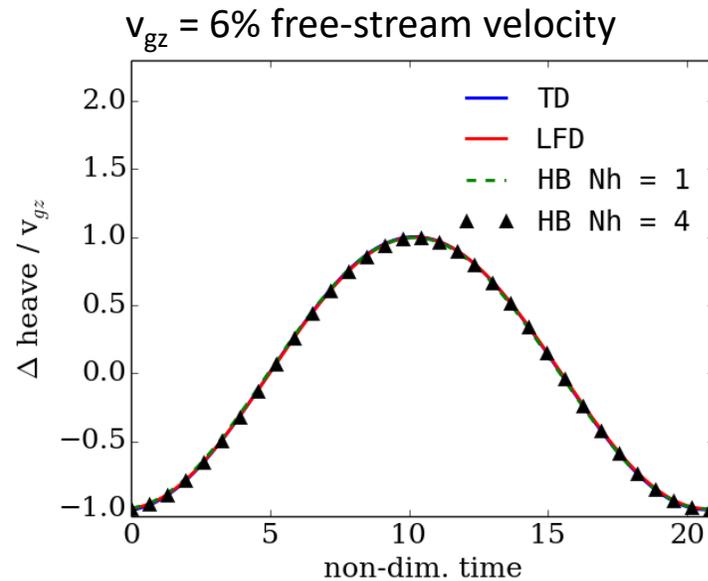
**Moment  
response**



## HB4Gust with FSI

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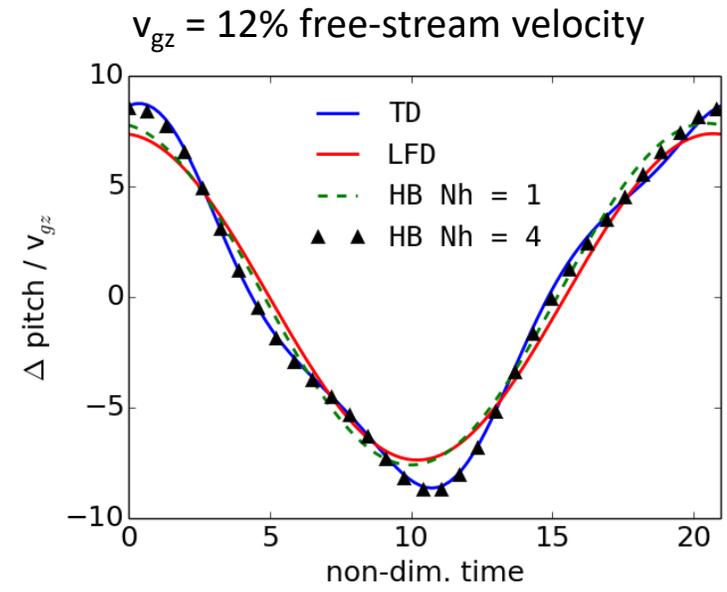
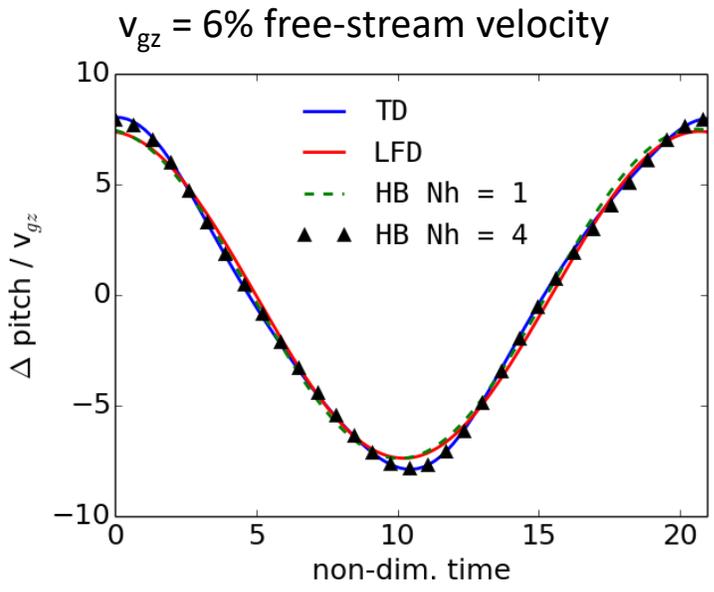
Heave response



# HB4Gust with FSI

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Pitch response



## Conclusion

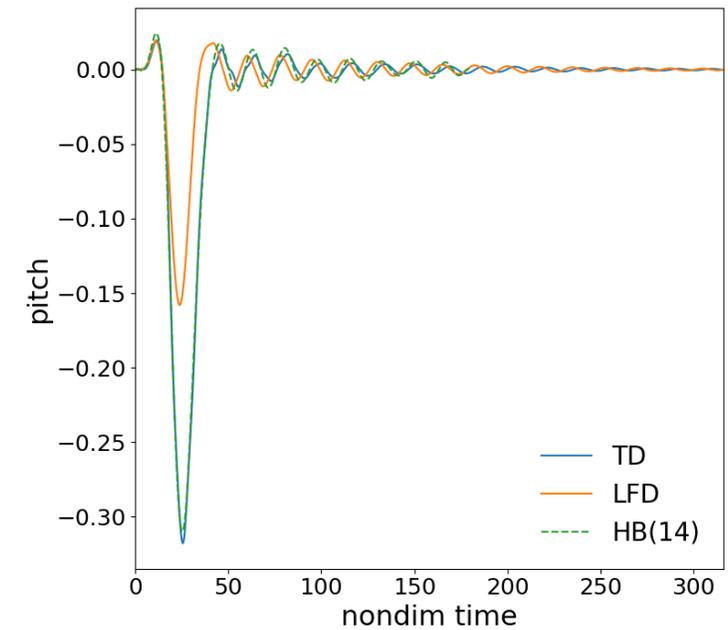
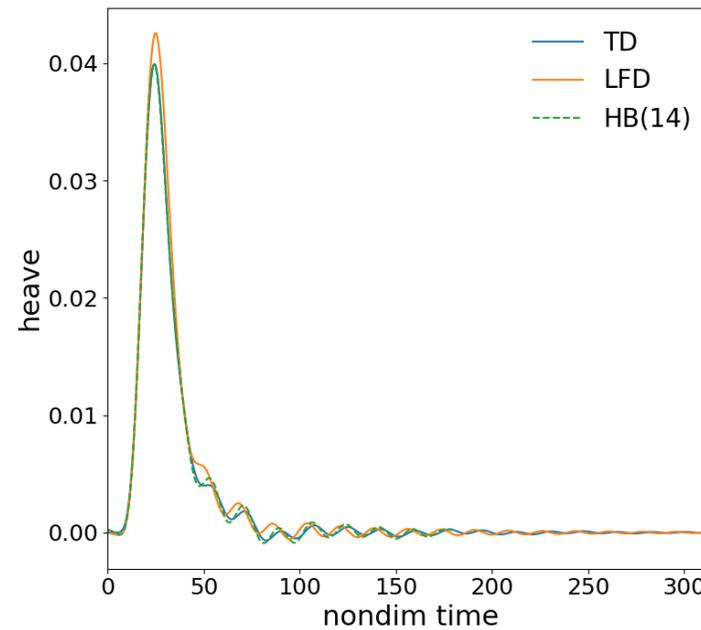
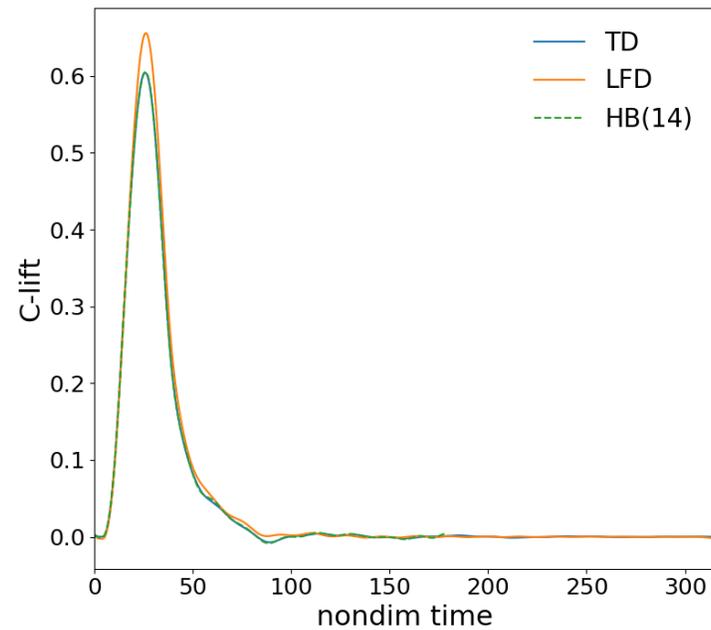
- LFD and HB solver extended to compute response of fluid-structure coupled systems due to gust encounter
- Demonstrated for sinusoidal gusts
  - Good agreement between HB(4) and TD reference
  - Lift and heave response dynamically linear → LFD sufficient
  - Contributions of higher harmonics for moment and pitch response
  - Nonlinearities captured well by HB method

### Future steps:

- Application to 1-cos gusts
- Apply symbiotic approach of HB-LFD to coupled system

## First results: 1-cos gust

- 1-cos response of longest gust:  $L_g = 35.5$ ,  $v_{gz} = 6.6\%$
- Lift and heave response is over-predicted by LFD while pitch response is under-predicted
- HB(14)(!) improves the prediction at the peak



**Thank you!**



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